NOTICE
Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

<table>
<thead>
<tr>
<th>Aircraft Make</th>
<th>Aircraft Model</th>
<th>Serial Number</th>
<th>Reg. Number</th>
</tr>
</thead>
<tbody>
<tr>
<td>Taylorcraft</td>
<td>BL-65</td>
<td>2082</td>
<td>N27440</td>
</tr>
</tbody>
</table>

1. Installed Garmin GTX 320 Transponder P/N 010-00135-00 in aircraft as per Garmin install manual #190-00133-01 Rev. H dated May 14, 1999.

2. Installed TCI SSD 120-30A TSO (TSO C88a) blind/encoding altimeter (to 30,000 ft.) using TCI installation manual #930004 dated July 1998.

3. Transponder/Altimeter integration system tested and inspected in accordance with 91.413 and FAR 43 appendices E and F.

4. Electrical load analysis computed: does not exceed 80% of the charging system output.

5. A/C was tested for EMI: None found

6. The alteration was also performed in accordance with AC43.13-1B Chapter 11, AC43.13-2A Chapters 2 & 3, and common industry standards.


**Continued Airworthiness**
Servicing and maintenance of this installation shall be accomplished in accordance with sound maintenance practices, FAR Part 43, and/or other data acceptable to the Administrator.

END

☐ Additional Sheets Are Attached

*U.S.GPO:1994-568-012/00019